# **MOSES** Instrument and Data Inversion

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## **Outline** — Instrument



- *MOSES* instrument concept
- Rocket instrument
- Benefits and costs
- Design approach for *Solar Orbiter*
- Specific design possibilities
- Conclusions





## MOSES—Multi-Order Solar EUV Spectrograph









- 1. Flight hardware complete—February, 2004
- 2. Calibration, I&T—Spring, 2004
- 3. Flight-Summer, 2004



## **Benefits and Costs**



#### Benefits:

- Speed (100 1000× comparable slit spectrograph)
- Simultaneity (no scanning in space or spectrum)

Costs:

- Detector real estate
- Inversion





- 1. Establish constraints (telemetry, volume, mass)
- 2. Choose lines and parameters to measure
- 3. Choose orders and dispersion
- 4. Verify inversion
- 5. Iterate & optimize



## **Design Possibilities**



- Gregorian layout
  - Magnification (compactness)
  - Field stop (edge effects, spectral selectivity)
  - TVLS gratings (Poletto & Thomas 2003)
- Multiple orders per detector
  - More spectral parameters, less ambiguity
  - Trade off: more detectors, or multiple orders per detector Example:  $4 k \times 4 k$  detector, 75 km pixels, 4 orders, 150,000 km FOV (leaving 96 pixel margin between orders).







- New capability: simultaneous measurement of spectral parameters in a narrow band, over a large 2D field of view.
- Propose specific observables as a starting point for design (Thanks Ben!)



## Inversion—Outline



- Inversion problems in solar physics
- Tomography in hybrid space  $(x, \lambda)$
- Fourier backprojection
- Pixon reconstruction
- Discussion
- Conclusions
- Next Steps







Inversion problems are endemic to remote sensing, and to solar physics in particular:

- MOSES
- Differential emission measure
- Coded-aperture imagers (HXT, RHESSI)
- Magnetography

For example, magnetography requires simplifying assumptions about the spectrum in a pixel.



## **Inversion** = **Tomography**







## **Fourier Backprojection**



- Each projection corresponds to a slice in the Fourier transform of  $I(x, \lambda)$ .
- Include horizonal projection,  $I_{\infty}(\lambda) = \int I(x, \lambda) dx$
- Iterate to enforce  $I(x, \lambda) > 0.$





**Characterizing**  $I_{\infty}(\lambda)$ 



The inversion relies on knowing  $I_{\infty}(\lambda)$ , which is not measured directly.

- HeII line center can be derived from the data (and the 3) orders co-aligned) by cross-correlation.
- He II line width can be inferred in at least two ways:
  - Knowledge of typical solar values plus instrumental linewidth.
  - Convergence of iteration for non-negativity.
- Si XI contribution does not significantly influence results for He II profile.



#### **Fourier Results**









## **Pixon Reconstruction**



- Incomplete fuzzy Pixon basis
- $\chi^2$  goodness of fit
- Fit the data using the least information.
- Include horizonal projection  $I_\infty(\lambda)$





#### **Pixon Results**











Differences between true and reconstructed line profile parameters. Line center and linewidth errors are in pixels. For each parameter, a mean offset and RMS error are given. One *MOSES* pixel is approximately 20 mÅ (20 km/s at 304 Å).

	center (median)		width (quartiles 1-3)	
Algorithm	mean	RMS	mean	RMS
Fourier	-0.01	0.32	-0.19	0.97
Pixon	-0.04	0.36	-0.04	0.59



### Slit spectrograph





Centroiding error due to Poisson noise only.



## **MOSES** backprojection





The glass is half full: if error is systematic, perhaps it can be corrected.







- Inversion demonstrated by both Fourier and Pixon methods.
- HeII is recovered, but not the weaker Si XI.
- Doppler shift reconstructed to  $\frac{1}{3}$ -pixel accuracy.
- Linewidth is reconstructed best by pixons (0.6 pixel).
- Exploring several avenues to improve performance.

A paper is in press for *Proc. SPIE*. Preprints available upon request.



**Next Steps** 



- Complete the Pixon algorithm (Fox & Metcalf).
- Parametric forward modeling (Fox).
- Understand and correct systematic errors in reconstruction.
- Quantify benefits of adding more orders (Theissen).
- Simulation/validation of specific wavelength options for *Orbiter* (Dobke).



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